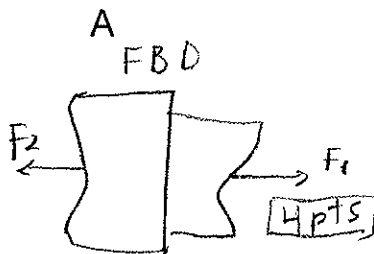
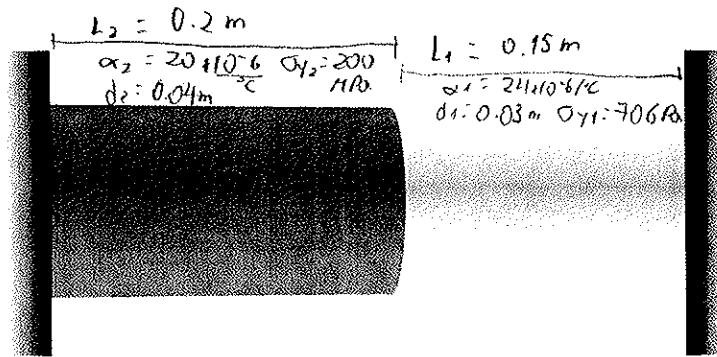


October 1, 2008

Instructor \_\_\_\_\_

**PROBLEM #2 (33 points)**

An assembly of two cylindrical rods (AB) and (BC) is restrained at both ends. Rod AB (length  $L = 200$  mm, diameter 40 mm) is made of brass (modulus of elasticity  $E = 100$  GPa, coefficient of thermal expansion  $\alpha = 20 \times 10^{-6} / ^\circ C$  and yield strength  $\sigma_y = 200$  MPa). Rod BC (length 150 mm, diameter 30 mm) is made of aluminum (modulus of elasticity  $E = 70$  GPa, coefficient of thermal expansion  $\alpha = 24 \times 10^{-6} / ^\circ C$  and yield strength  $\sigma_y = 80$  MPa). The system is heated by  $100^\circ C$ . Determine if the system is safe or will fail by plastic deformation.



$$\sum F_x = 0$$

$$-F_2 + F_1 = 0$$

$$F_2 = F_1$$

4 pts

B

Geometry of Deformation:

$$e_1 + e_2 = 0$$

$$e_1 = -e_2 \quad \boxed{8 \text{ pts}}$$

$$e_1 = \frac{F_1 L_1}{E_1 A_1} + L_1 \alpha_1 \Delta T$$

$$= \frac{F_1 (0.15 \text{ m})}{70 \times 10^9 \text{ Pa} \cdot \pi (0.015 \text{ m})^2} + 0.15 \text{ m} (24 \times 10^{-6} / ^\circ C) (100^\circ C)$$

$$= 3.031 \times 10^{-3} F_1 + 0.00036 \text{ m}$$

$$e_2 = \frac{F_2 L_2}{E_2 A_2} + L_2 \alpha_2 \Delta T \quad \boxed{8 \text{ pts}}$$

$$= \frac{F_2 (0.2 \text{ m})}{100 \times 10^9 \text{ Pa} \cdot \pi (0.02)^2} + 0.2 \text{ m} (20 \times 10^{-6} / ^\circ C) (100^\circ C)$$

$$= 1.147 \times 10^{-3} F_2 + 0.0004 \text{ m}$$

using  $e_1 = -e_2$  and  $F_2 = F_1$

$$3.031 \times 10^{-3} F_1 + 0.00036 = -1.147 \times 10^{-3} F_1 - 0.0004$$

$$4.172 \times 10^{-3} F_1 = -0.00076$$

$$F_1 = -182.166 \text{ kN} \quad \boxed{5 \text{ pts}}$$

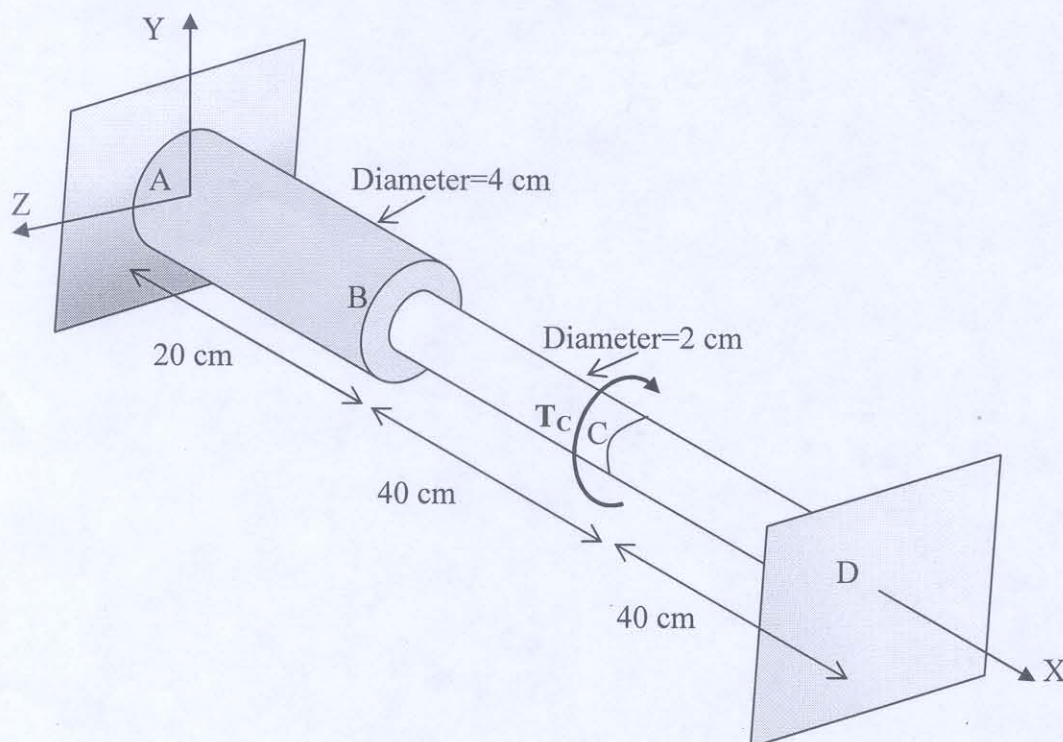
$$F_2 = -182.166 \text{ kN}$$

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**PROBLEM #4 (33 points)**

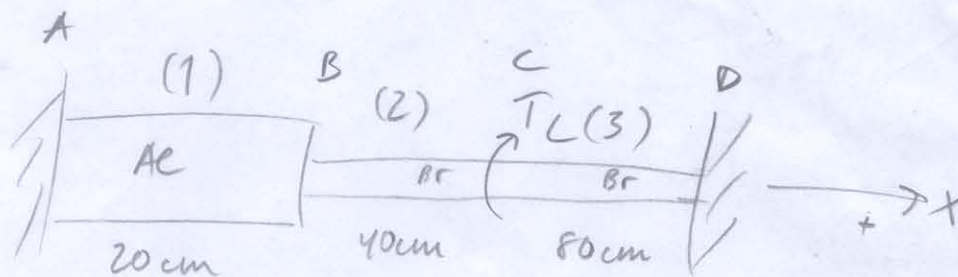
The stepped rod shown below is made of Aluminum and bronze (shear modulus of aluminum  $G_{al}=25$  GPa and shear modulus of bronze  $G_{br}=20$  GPa) and is fixed to the walls at A and D. The aluminum section is 20 cm long and has a diameter of 4 cm. The bronze section is 80 cm long and has a diameter of 2 cm. An external torque  $T_C$  is applied as shown. The yield stress of the aluminum and bronze are  $\tau_Y = 20$  MPa and 10 MPa respectively. Determine the maximum allowable applied torque at C,  $T_C$ ?



Exam #1

Prob. 2

FBD



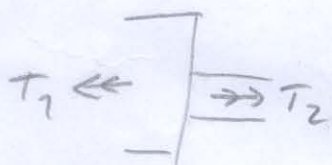
A:



$$T_1 - T_A = 0$$

$$T_1 = T_A$$

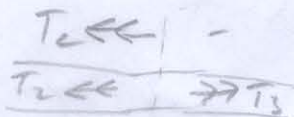
B:



$$T_2 - T_1 = 0$$

$$T_1 = T_2$$

C:



$$T_3 - T_2 - T_C = 0$$

$$T_C = T_3 - T_2$$

D:



$$T_D - T_3 = 0$$

$$T_3 = T_D$$

8 pts

$$I_{PAE} = \frac{\pi d_{AE}^4}{32} = 8 \cdot 10^{-8} \pi \text{ m}^4$$

$$I_{PBr} = \frac{\pi d_{Br}^4}{32} = 5 \cdot 10^{-9} \pi \text{ m}^4$$

2 pts

$$T_1 = \frac{\tau_{AE} \cdot I_{PAE}}{r_{AE}} = \frac{20 \cdot 10^6 \cdot 8 \cdot 10^{-8} \pi}{0,02} = 80 \pi \text{ Nm}$$

$$T_2 = \frac{\tau_{Br} \cdot I_{PBr}}{r_{Br}} = \frac{10 \cdot 10^6 \cdot 5 \cdot 10^{-9} \pi}{0,01} = 5 \pi \text{ Nm}$$

9 pts

By FBD  $T_1 = T_2 = 5 \pi \text{ Nm}$

$$\varphi_1 = \frac{T_1 L_{AE}}{G_{AE} I_{PAE}} = \frac{5 \pi \cdot 0,2}{25 \cdot 10^9 \cdot 8 \cdot 10^{-8} \pi} = 0,005$$

$$\varphi_2 = \frac{T_2 L_{Br2}}{G_{Br} \cdot I_{PBr}} = \frac{5 \pi \cdot 0,4}{20 \cdot 10^9 \cdot 5 \cdot 10^{-9} \pi} = 0,02$$

$$\varphi_3 = \frac{T_3 L_{Br3}}{G_{Br} \cdot I_{PBr}} = \frac{T_3 \cdot 0,8}{20 \cdot 10^9 \cdot 5 \cdot 10^{-9} \pi} = \frac{0,008}{\pi} T_3$$

12 pts

Deformation geometry

$$\varphi_1 + \varphi_2 + \varphi_3 = 0$$

$$\Rightarrow 0,005 + 0,02 + \frac{0,008}{\pi} T_3 = 0$$

$$T_3 = -2,5625 \pi \text{ Nm}$$

$$T_C = T_3 - T_2 = -2,5625 \pi - 5 \pi = -7,5625 \pi = -23,7583 \text{ Nm}$$

3 pts

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Check if section 1 or 2 fails:

$$\sigma = \frac{F}{A} \quad \boxed{2 \text{ pts}}$$

$$\sigma_1 = \frac{F_1}{A_1} = \frac{182166 \text{ N}}{\pi (0.015)^2} = 2570.712 \text{ MPa} > 80 \text{ MPa} \quad \text{it fails!}$$

$$\sigma_2 = \frac{F_2}{A_2} = \frac{182166 \text{ N}}{\pi (0.02)^2} = 144.96 \text{ MPa} < 200 \text{ MPa} \quad \text{OK!}$$

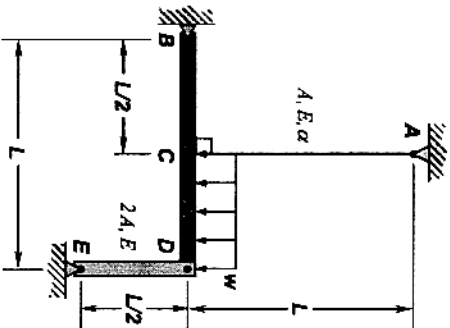
System fails since  $\sigma_1 > \sigma_{y1}$  2 pts

total = 33 Points

**PROBLEM #5 (33 points)**

The rigid beam BCD shown below is supported by a frictionless pin at point B, vertical thin wire AC at point C, and bar DE at point D. Assume the following:

- Prior to loading, beam BCD is horizontal
  - Prior to loading, there is no stress in wire AC or rod DE
  - Weight of beam BCD is negligible
  - There is very small displacement at point D
  - Wire AC has cross section area = A, modulus of elasticity, E
  - Bar DE has cross section area = 2A, and modulus of elasticity, E
- The uniformly distributed load with intensity w per unit length is applied across section CD. Develop expressions for the axial forces in wire AC and bar DE.



**10pts** I) EAULT FBD BCD



$$+\circlearrowleft (\sum M)_B = 0$$

$$-F_{AC} (L/2) + wL/2 (L/4) + F_{DE} (L) = 0$$

$$4F_{AC} - 8F_{DE} = 3wL \quad \text{E@1}$$

**10pts** II) FORCE - DEFORM BEHAVIOR

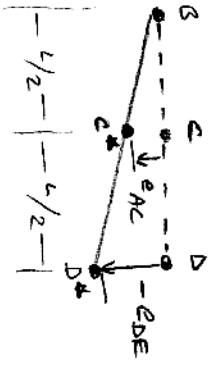
$$\epsilon_{AC} = \left( \frac{F_{AC} L}{AE} \right) ; \left. \begin{matrix} \Delta_{AC} = L \\ \Delta_{AC} = A \end{matrix} \right\} = f$$

$$\epsilon_{AC} = \frac{F_{AC} L}{AE} = f \quad \text{E@2}$$

$$\epsilon_{DE} = \left( \frac{F_{DE} L}{2AE} \right) ; \left. \begin{matrix} \Delta_{DE} = L/2 \\ \Delta_{DE} = 2A \end{matrix} \right\} = \frac{f}{4}$$

$$\epsilon_{DE} = \frac{F_{DE} L}{4AE} = \frac{f}{4} \quad \text{E@3}$$

**10pts** III) DEFORM GEOM (i.e. compatibility)



$$2\epsilon_{AC} + \epsilon_{DE} = 0 \quad \text{E@4}$$

• COMBINE E@1-4

$$2(F_{AC}) + \frac{f}{4} F_{DE} = 0 \Rightarrow 8F_{AC} + F_{DE} = 0 \quad \text{E@5}$$

• SOLVE E@1 / 4'S SIMULTANEOUSLY

$$-2[4F_{AC} - 8F_{DE} = 3wL]$$

$$\frac{8[2F_{AC} + F_{DE} = 0]}{(4+84)F_{AC} = 3wL} \Rightarrow F_{AC} = \frac{3}{68} wL$$

$$\Rightarrow (16+1)F_{DE} = -6wL \Rightarrow F_{DE} = -\frac{6}{17} wL$$